AEROSPACE MEDICAL RESEARCH LAB WRIGHT-PATTERSON AFB OH F/6 1/2 USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK. VOLUME 145. F-106B I--ETC(U) JUL 79 H K HILLE AMRL-TR-75-50-VOL-145 NL AD-A081 847 UNCLASSIFIED END 4 80 DTIC

SECURITY CLASSIFICATION OF THIS PAGE (When Date Entered) READ INSTRUCTIONS BEFORE COMPLETING FORM REPORT DOCUMENTATION PAGE 2. GOVT ACCESSION NO. 3. RECIPIENT'S CATALOG NUMBER AMRL-TR-75-50 TYPE OF REPORT & PERIOD COVERED SAE BIOENVIRONMENTAL NOISE DATA HANDBOOK Volume 1450 of a Series PERFORMING ORG. REPORT NUMBER S. CONTRACT OR GRANT NUMBER(s) AUTHOR(#) Tedhnical reptis Harald K./Hille 9. PERFORMING ORGANIZATION NAME AND ADDRESS PROGRAM ELEMENT, PROJECT, TASK AREA & WORK JUNIT NUMBERS Aerospace Medical Research Laboratory Aerospace Medical Division, Air Force Systems Command, Wright-Patterson AFB OH 62202F 11. CONTROLLING OFFICE NAME AND ADDRESS 12. REPORT DAT Jul Same as above 14. MONITORING AGENCY NAME & ADDRESS(If different from Controlling Office) 15. SECURITY CLASS. (of this report) Unclassified 15a DECLASSIFICATION DOWNGRADING 16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited. 17. DISTRIBUTION STATEMENT (of the obstract entered in Black 20. if different from Report) 18. SUPPLEMENTARY NOTES B 19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Noise F-106B Aircraft Noise Environments Bioenvironmental Noise In-Flight Crew Noise 20. ABSTRACT (Continue on reverse side if necessary and identify by block number) The F-106B is a USAF all-weather fighter. This report provides measured data defining the bioacoustic environments at the pilot's location inside this aircraft for 22 flight conditions. Data are reported for one location in a wide variety of physical and psychoacoustic measures: overall and band sound pressure levels, C-weighted and A-weighted sound levels, preferred speech interference level, perceived noise level, and limiting times for

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total daily exposure of personnel with and without standard Air Force ear protectors. Refer to Volume 1 of this handbook, "USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application," AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc.

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PREFACE

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 723108, Crew Safety In Operational Noise Environments.

The author acknowledges the efforts of Mr. John N. Cole who established the data analysis requirements, Mr. Henry Mohlman and Mr. Fred Lampley of the University of Dayton who assisted in the mechanics of data processing and Mrs. Norma Peachey who typed this report and prepared it for publication.

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INTRODUCTION

The F-106B is a USAF all-weather fighter manufactured by the Convair Division of General Dynamics. This aircraft is powered by one J75-P-11 turbojet engine rated at 24,500 lbs. maximum takeoff thrust with afterburner. The engine is manufactured by the United Aircraft Corporation, Pratt and Whitney Aircraft Division.

This volume provides measured data defining the bioacoustic environments produced inside the aircraft. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with operations of the F-106B aircraft.

This volume is one of a series published by the Aerospace Medical Research Laboratory (AMRL) under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and ground support equipment. The far-field, community-type, noise data in the handbook describe the noise produced during ground operations of aircraft, ground support equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. Refer to Volume 1 reference. For such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., in-flight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published, and is available upon request from AMRI/BBE, Wright-Patterson AFB, OH 45433 Organisations on the distribution list for the handbook will automatically receive a copy of the updated in the as it is generated.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; AUTOVON 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

1. Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application, AMRL-TR-75-50 (1), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

IN-FLIGHT NOISE

Measurements

All noise measurements were made on-board a F-106B aircraft during typical speed, altitude, and flight maneuver conditions. These levels describe the standard F-106B environments but may not be representative of those levels encountered if the aircraft has been configured differently (e.g., major equipment or structural changes).

Acoustic measurements were made inside the cockpit at the pilot's location. Table 1 lists the measurement location and test conditions as numeric/alphabetic designators which are used on the data pages. The designator 1/A means measurement location 1 and test condition A, etc.

The microphone was attached to the pilot's helmet by means of a lightweight boom. This arrangement enabled adjustment of the microphone close to the ear level at a distance of 0.1 meter with its diaphragm parallel and facing away from the helmet's surface. In the analysis, microphone corrections for random incidence were applied to the overall systems response. The recorded samples were analyzed using a four or eight second integration time to obtain a power-averaged level which effectively smooths out short duration fluctuations and best describes the exposure.

Results

The measured data presented in Table 2 define the sound pressure levels (SPL) produced inside the F-106B aircraft at the specified location. This table includes the overall, ½ octave band, and octave band levels. From these data, C-weighted and A-weighted sound levels, maximum permissible time for one exposure per day (AFR 161-35) with and without standard Air Force ear protectors, preferred speech interference level, and perceived noise level are calculated and presented in Table 3. These measures are widely used to assess the effects of noise on personnel and their performance.

TABLE 1

MEASUREMENT LOCATIONS AND TEST CONDITIONS F-106B, Tyndall AFB FL, 8 June 78 Tail #904

Location	Position	Height above Deck
1	Pilot, Front Seat	Seated Head Level
Condition	Description	
Α	Ground Runup - Engine Idle - Car	nopy Open
В	Ground Runup - Engine Idle - Car	
\mathbf{C}	Taxi - Engine Idle - Canopy Open	
D	Taxi - Engine Idle - Canopy Close	d
\mathbf{E}	Takeoff	
\mathbf{F}	Cruise - 10,000' PA - 325 KIAS	
G	Cruise - 14,000' PA - 340 KIAS	
H	Cruise - 25,000' PA - 375 KIAS, 0	0.95M
I	Intercept - 27,000' PA - 375 KIAS	}
J	Reattack - 31,000' PA - 375 KIAS	
K	Cruise - 28,000' PA - Engine A/B,	300 KIAS
L	Cruise - 27,000' PA, 0.95M, IR See	eker Head Up
M	Cruise - 31,000' PA, 1M, IR Seeke	r Head Up
N	Descent To 21,000' PA - 385 KIAS	
O	Descent To 21,000', Speed Brakes	Out – 385 KIAS
P	Cruise - 9,000' - 275 KIAS	
Q	Cruise - 3,000' - 275 KIAS, IR See	eker Head Up
R	Cruise - 3,000' - 275 KIAS, IR Se	eeker Head Down
S	Initial	
T	Gear Down – 225 KIAS	
U	Touchdown	
V	Roll – 100 KIAS	

TABLE : MEASURED SOUND F	OUND PRESSURE	E LEVEL	L (03)) I DENTIFICATION:
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04	86	80	91	91	102	40	82	79	11	11	80
50	66	75	96	48	96	82	7.7	73	7.4	7.7	16
63	97	92	85	8	113	7.8	75	22	69	20	73
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101	%	93	93	97	105	8 0 (97	97	& ! 6	97	100
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500	86	48	95	82	36	96	96	93	96	83	de 5
630	32	79	93	11	68	97	96	91	92	81	78
008	66	80	46	4	87	91	83	8	40	82	3
1000	100	81	701	78	90	86	82	8	92	85	200
1250	103	8	105	6.	9	67	90	80	. † (9	,
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5000	66	72	66	69	47	75	92	82	78	77	79
6306	100	12	101	69	2	77	22	9	92	86	6
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OVERALL	112	97	112	101	115	165	108	104	110	66	102
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LEVEL CORRECTED TO REMOVE BACKGROUNDZELECTRONIC NOISE.

TABLE: MEASURED SOUND P	PRESSURE	RE LEVEL	L (03)							} !) I DENTIFICATION:
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	. .) PAGE F1
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004	66		98	85	96	46	104	35	48	82	83
206	96		95	82	5 6	98	9 6	93	96	83	85
630	95		93	77	68	4	96	91	92	81	700
800	66		97	62	87	91	68	96	97	82	***
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900	8		86	99	±	92	92	79	78	4	80
000	32		95	29	73	92	78	90	64	80	82
12500	93		91	63	69	4.	9,	78	78	49	80
OVERALL	112	46	112	101	115	165	108	104	190	66	102

LEVEL CORRECTED TO REMOVE BACKGROUNDZELECTRONIC NOISE.

(TABLE: MEASURED SOUND PRES	SSU	RE LEVE	r (03)			i ! !	; 1 1 1	; ; ;	† † †) IDENTIFI) OMEGA	CATION:
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C IN-FLIGHT CREW NOISE) PAGE F	٥
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0008	82		73	92	7.	73	2	7.	7.4	73	69	
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(OVERALL	114	112	95	66	66	106	136	101	120	113	105	•
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LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

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(F-106B AIRCRAFT (IN-FLIGHT CREW NOISE	ш										23 JUL 79) } PAGE J1)
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(125	66	76	95	86	107	160	100	100	96	96	100
(250	66		96	88	101	98	104	93	87	88	86
200	102		100	87	66	161	105	26	93	87	0 %
1000	105		107	83	91	93	91	46	39	86	88
5000	103		105	82	82	89	88	35	96	82	85
0004	104		105	79	81	83	94	83	48	82	84
0000	1 02		103	72	62	81	82	85	36	88	06
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500	104	105	96	37	88	66	93	92	66	5	06
1000	8	66	80	88	98	93	68	96	91	87	85
2008	æ	95	81	87	3	86	96	96	87	†	60
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HGU-24/P HELYET WITH H-154	54	;	i	•	;	•	1				ļ
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HGU-2A/P HELMET WITH H-15	H-154(A)	5	9	1	1	7 4	3	707			r s
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F-136B AIRCRAFT											23 JUL 79
IN-FLIGHT CREW NOISE											PAGE H2
	1,1	£	1 / N	1,0	LOCATION/CONDITION	N/COND 1/0	ITION 1/3	1/5	7.1	1/U	1/ V
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